



Quantifying the Need for Advanced Computational Tools for Lunar Excavation Analysis

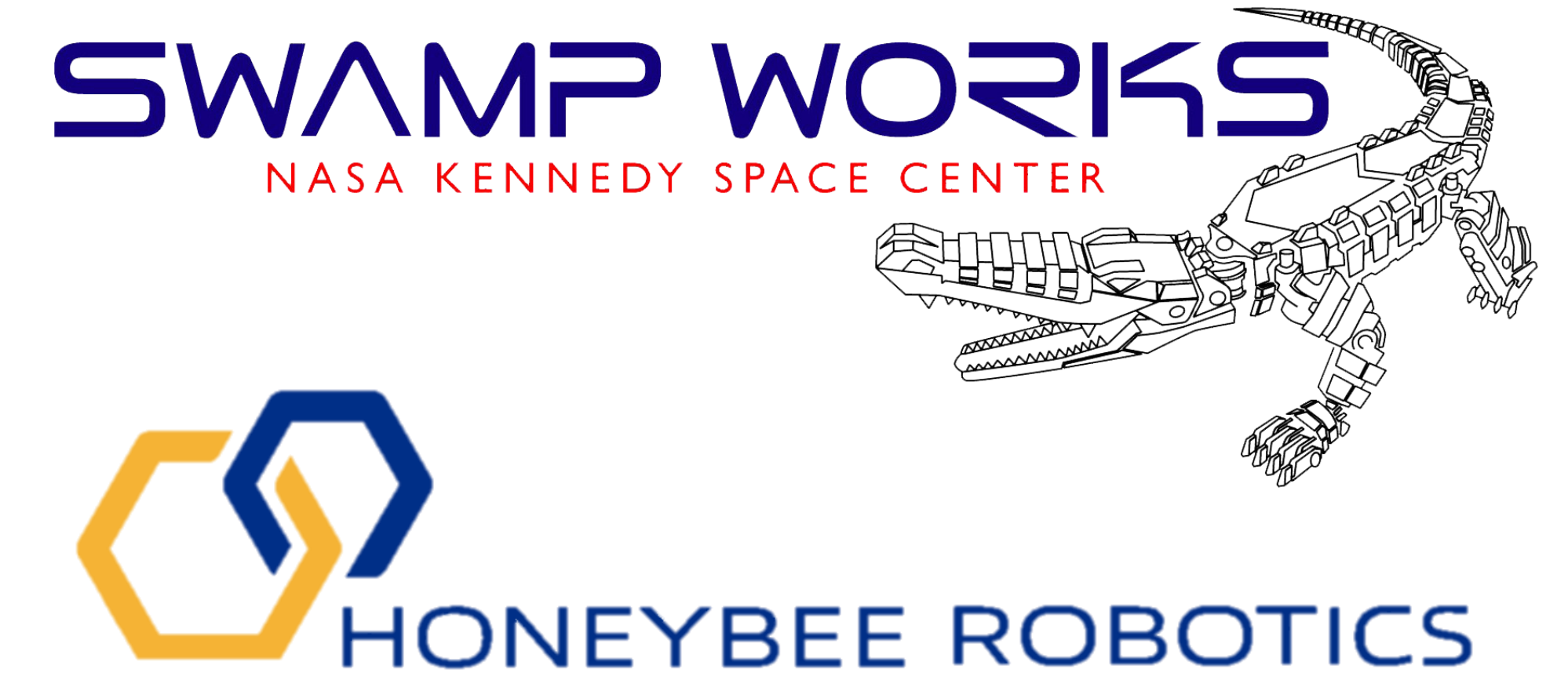
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Introduction

- Infrastructure development and ISRU on the lunar surface will require thousands of tons of lunar regolith → efficiency is key
- Excavation force reduction
 - Percussive [1] and vibratory [2]
 - Return on investment must be enough to justify the extra power and equipment complexity
- Computational models must enable comparison of different scoop, tool path, and force reduction methods [3], power requirements, and efficiency

Methods

- Excavation force and percussion motor power draw data from [1]; replica Surveyor III SMSS scoop, JSC-1A simulant
- Analytical Modeling
 - Static, medium RD, 70° rake angle, 5 mm/s speed
 - 2D Reece's Fundamental Equation of Earthmoving (FEE) [4]
 - Predict excavation forces based on known regolith/simulant properties (forward model)
 - Predict regolith/simulant properties from measured excavation forces (inverse model)
- Excavation Efficiency (force vs power)
 - Varying percussive BPM, high RD, 70° rake angle, 5 mm/s speed

2D Reece's Fundamental Equation of Earthmoving

$$F = w \left[cdN_c + \gamma d^2 N_\gamma + QdN_q \right]$$

$$N_c = \frac{1 + (\cot(\beta) \cot(\beta + \phi))}{[\cos(\rho + \delta) + (\sin(\rho + \delta) \cot(\beta + \phi))] \cot(\rho) + \cot(\beta)}$$

$$N_\gamma = \frac{2[\cos(\rho + \delta) + (\sin(\rho + \delta) \cot(\beta + \phi))] \cot(\rho) + \cot(\beta)}{2[\cos(\rho + \delta) + (\sin(\rho + \delta) \cot(\beta + \phi))] \cot(\rho) + \cot(\beta)}$$

$$N_q = \frac{\cot(\rho) + \cot(\beta)}{\cos(\rho + \delta) + (\sin(\rho + \delta) \cot(\beta + \phi))}$$

where F is excavation force, w is scoop width, d is scooping depth, γ is density, Q is surcharge load, ϕ is angle of internal friction, δ is angle of external friction, ρ is rake angle, and β is shear plane failure angle

Analytical Modeling

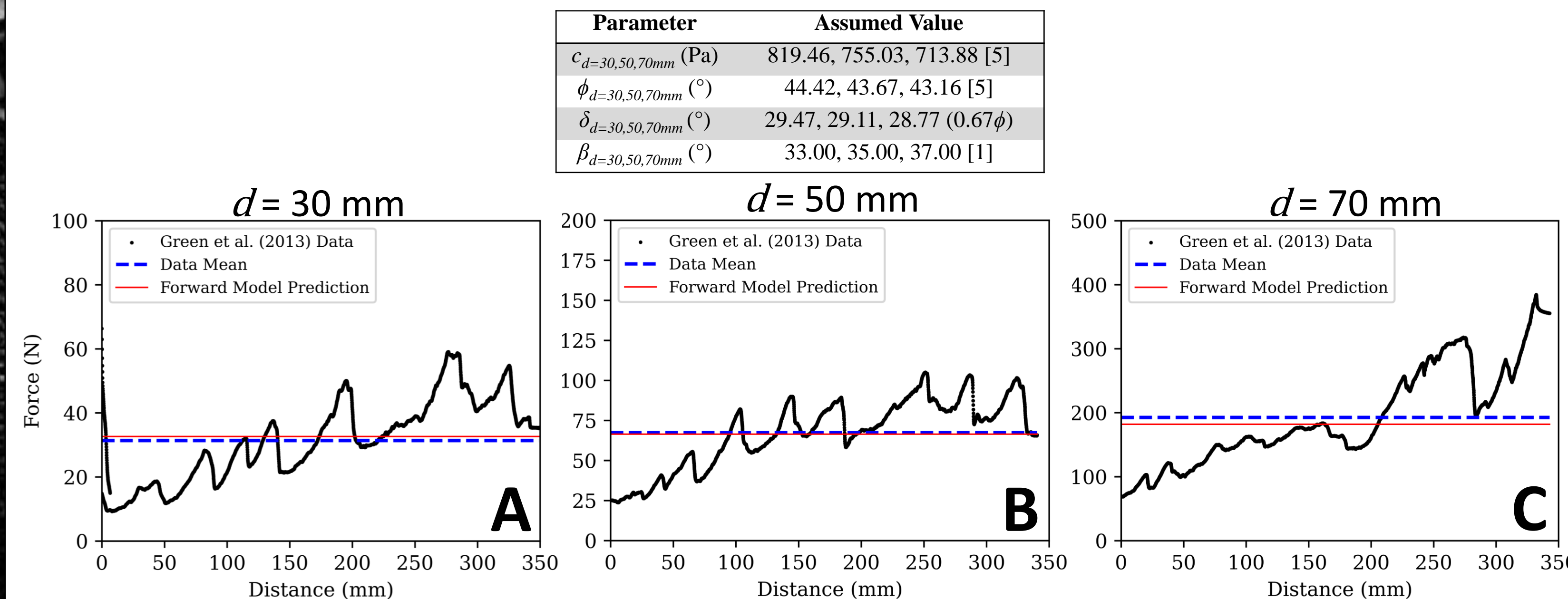


Figure 1. Forward model predictions of data from [1] for excavation speeds of 5 mm/s, rake angle of 70°, and depths of (A) 30 mm, (B) 50 mm, and (C) 70 mm using Reece's FEE and given input parameter values.

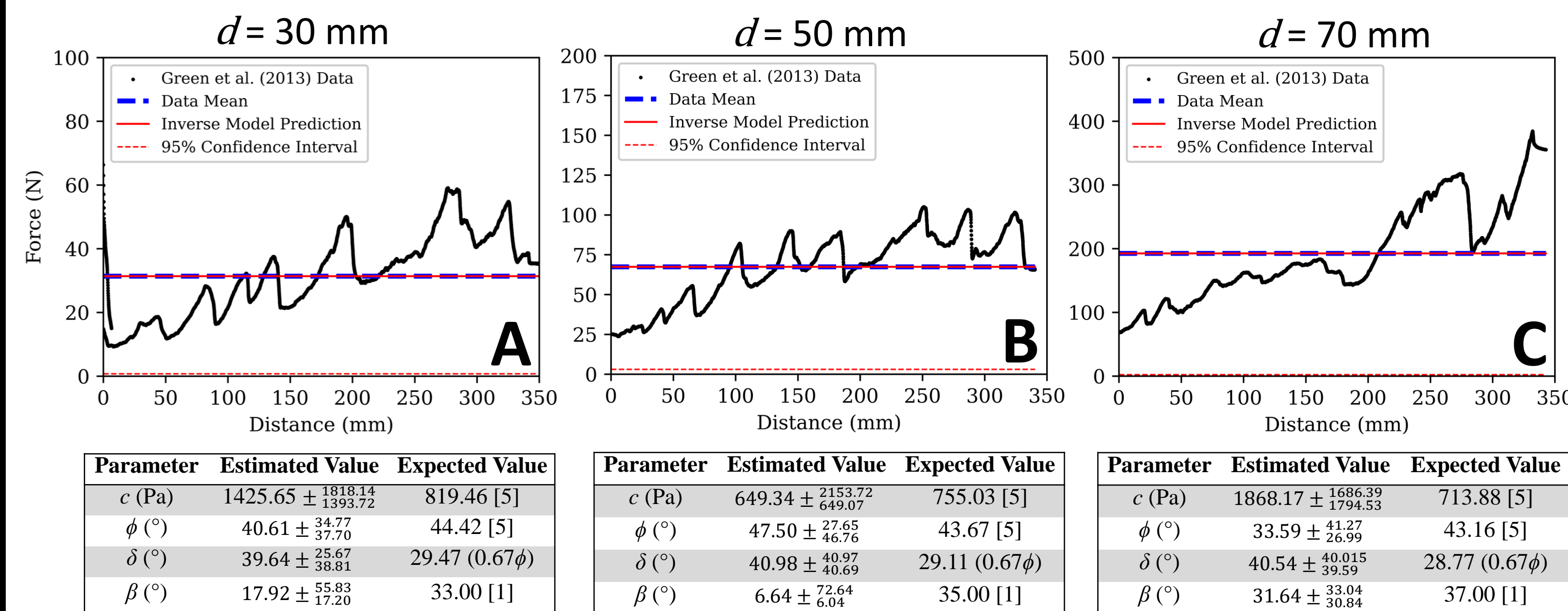


Figure 2. Inverse model predictions and uncertainties of model parameters based on data from [1] for excavation speeds of 5 mm/s, rake angle of 70°, and depths of (A) 30 mm, (B) 50 mm, and (C) 70 mm.

Metrics of Excavation Efficiency

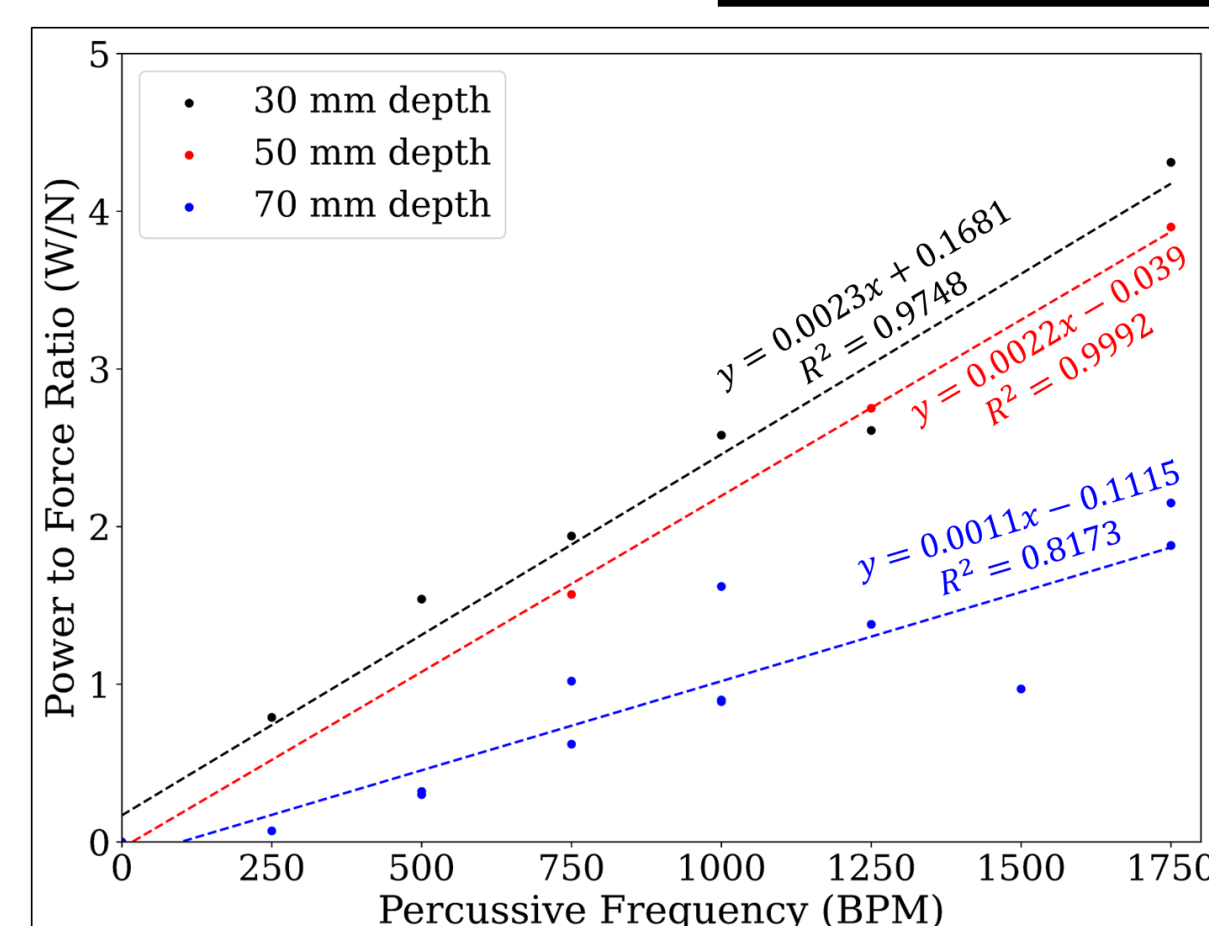


Figure 3. Linear fits to percussive frequency vs percussion motor power expended per unit excavation force during excavation of high relative density JSC-1A with a replica Surveyor III SMSS scoop at 30, 50, and 70 mm digging depths; data from [1].

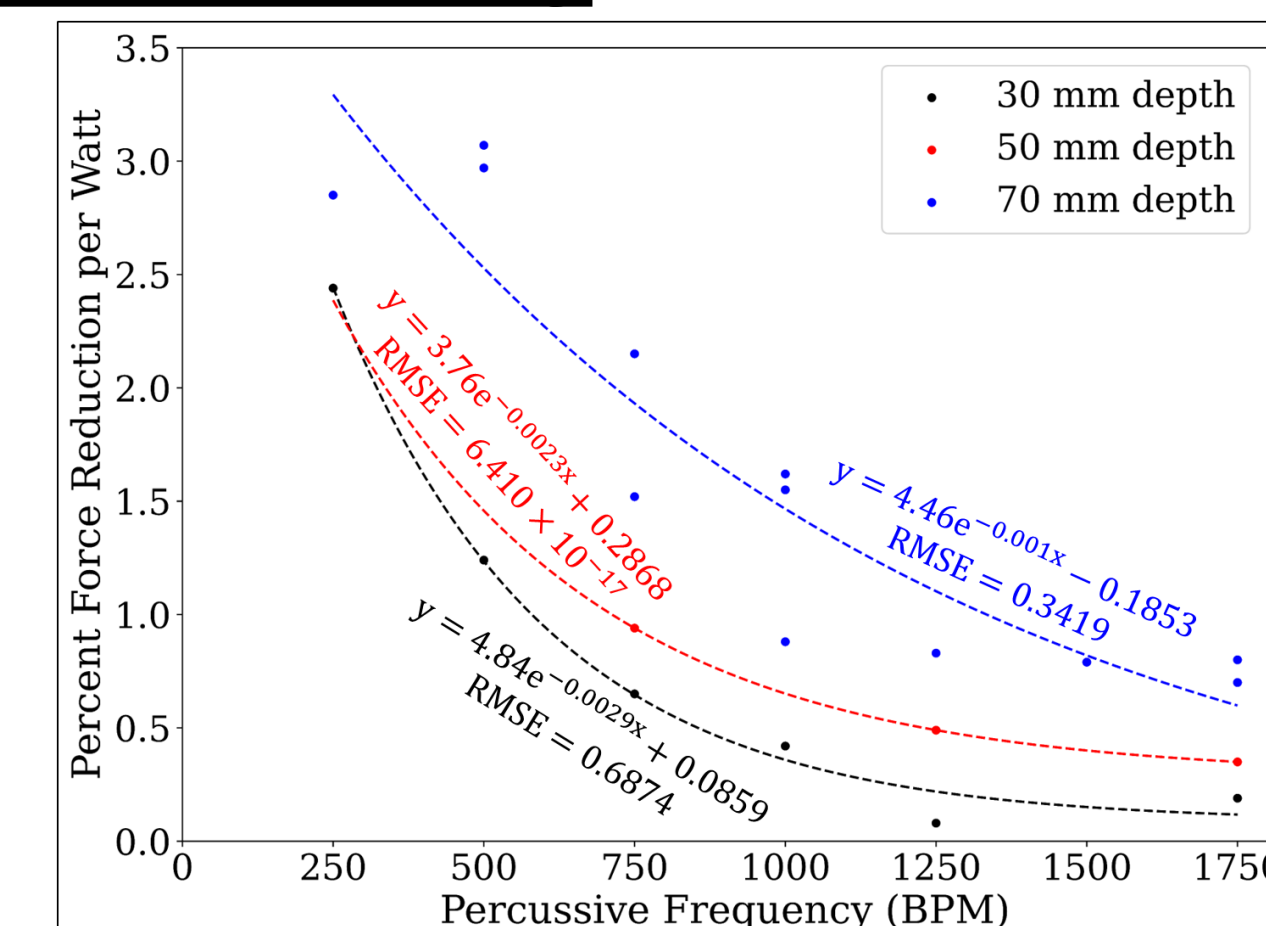


Figure 4. Exponential curve fits to percussive frequency vs percent force reduction per watt expended by percussion motor during excavation of high relative density JSC-1A with a replica Surveyor III SMSS scoop at 30, 50, and 70 mm digging depths; data from [1].

Discussion

- Standard analytical models are not suitable for lunar and planetary excavation analysis
 - Forward models offer good predictions; inverse modeling reveals ill-resolved parameter space → not reliable
 - Not able to be used for tool path development, equipment design, power budget analysis, etc.
- Efficiency analysis shows:
 - Power cost per unit force is linearly proportional to percussive frequency
 - Percent force reduction per unit power expended has diminishing returns → can be optimized
 - General trends expected to hold for other scenarios (regolith, equipment)

Conclusion

- More advanced computational models are needed for planetary excavation hardware design, CONOPS development, and power budget analysis
 - Subject of ongoing work

Acknowledgements

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References

- [1] Green et al. (2013), *J. Aero. Eng.* 26(1).
- [2] Rezich et al. (2021), *ASCE Earth and Space*.
- [3] Zacny et al. (2010), *ASCE Earth and Space*.
- [4] Reece, A. R. (1964), *Proc. Inst. Mech. Eng.* 1964-65.
- [5] Dotson et al. (2024), *Icarus* 411.